







# THERMO-MECHANICAL FATIGUE CRACK GROWTH IN ADVANCED AEROSPACE ALLOYS

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#### Introduction





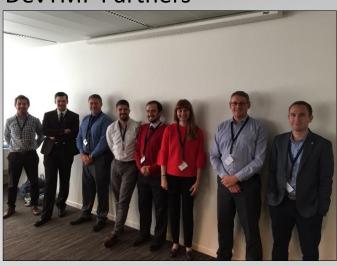
#### Swansea University Bay Campus



DevTMF. This project has received funding from the *European Union's Horizon 2020 research and innovation programme* and Joint Undertaking Clean Sky 2 under grant agreement No 686600.



#### **DevTMF Partners**



**Swansea** University, Wales. *Testing and analysis* 

**Nottingham** University, England.

Modelling and round robin testing

**Linkoping** University, Sweden.

Modelling and round robin testing

Rolls-Royce plc, UK.

Material and technical support

#### Introduction





- Swansea University Background in TMF
- TMF total life testing
- > TMFCG Test Development
- Crack tip heating investigations
- > TMFCG Test Results
- Phase angle effects
- Damage mechanisms

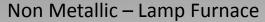


# **Background in TMF**





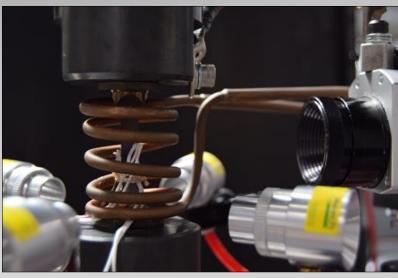
Strain Control - Induction



Fatigue Crack Propagation – Induction



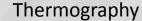
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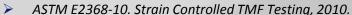


Thermocouple

Rear View

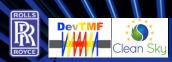
**Pyrometer** 





- ISO 12111:2011. Strain-controlled TMF Testing, 2011.
- **BAM.** CoP Force-Controlled TMF Testing, 2015.

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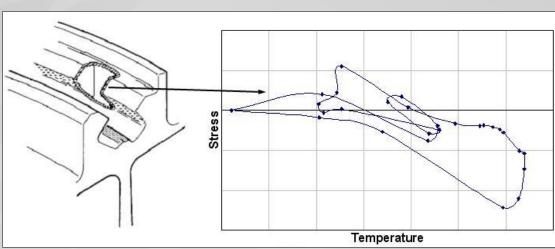
#### **Industrial Motivation**





- Increased turbine entry temperatures
- Thinner disc rims and advanced cooling systems leading to larger thermal gradients
- Complex loading regimes within the gas turbine leading to diverse phasing between temperature and strain
- Extrapolation of isothermal fatigue (IF) results to incorporate these effects show limited success
- Generation of TMF data is required to allow the development of lifing methodologies under <u>TMF</u> loading





# **Thermo-Mechanical Fatigue (TMF)**

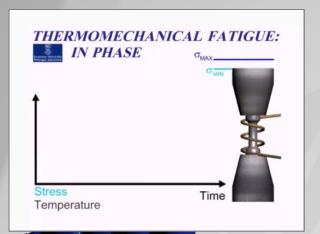


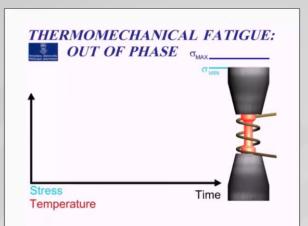


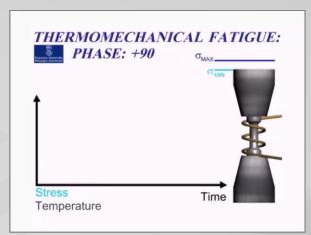
Diverse mechanisms are involved, Primarily . . .

# Fatigue Creep Oxidation

- TMF loading can be more damaging than isothermal fatigue at an equivalent T<sub>max</sub>
- Complex interaction within diverse phase angles between peak temperature and strain range





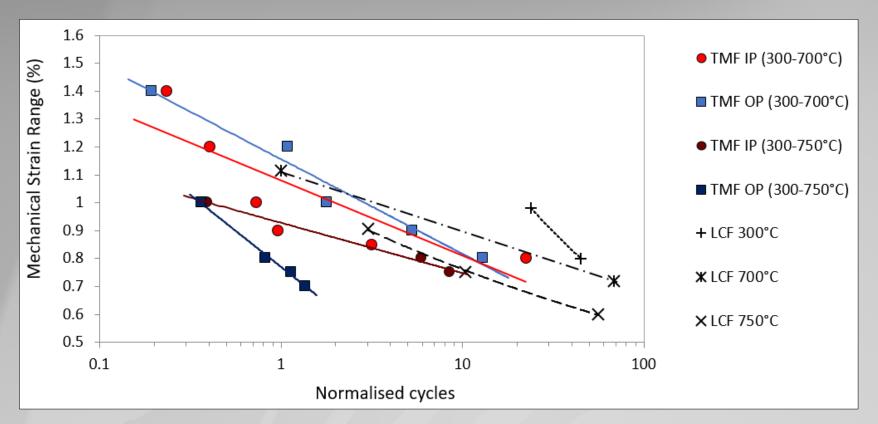




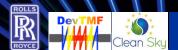
# **Effect of peak temperature**







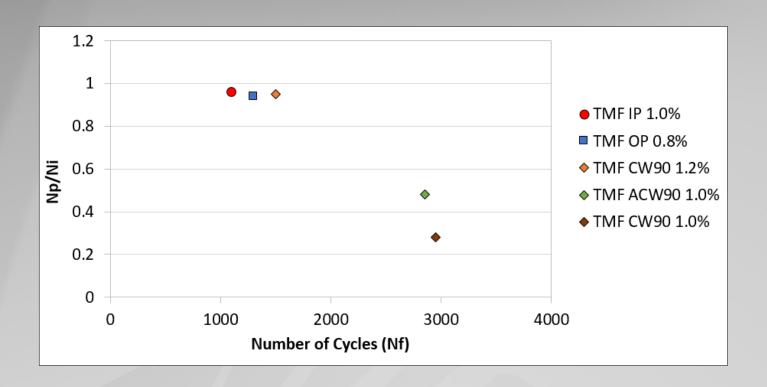
- > At 750°C OP data shows a significant decrease in TMF life.
- Likely to be due to increased oxidation effects
  - TMF lives consistently shorter than isothermal fatigue lives.



# **Crack propagation**





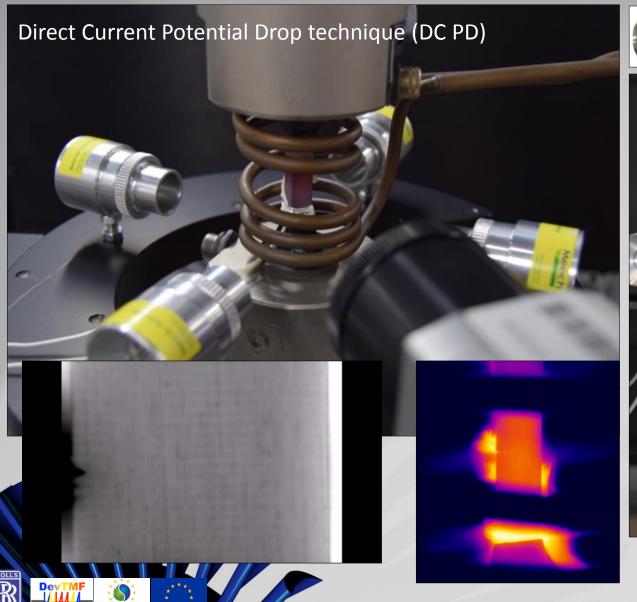


For fatigue lives that are less than 5000 cycles it is not appropriate to consider only crack initiation as the dominant factor in fatigue life.

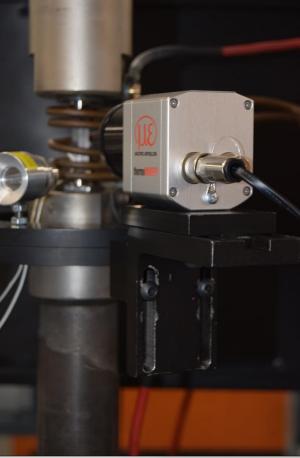
# **TMFCG Test Development**











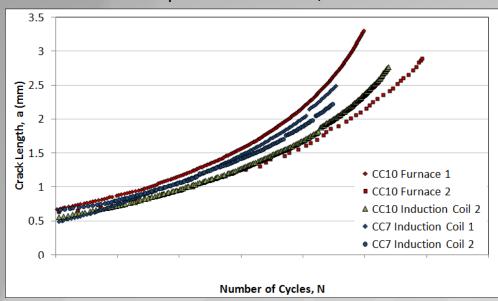
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# **Crack Tip Heating Investigations**

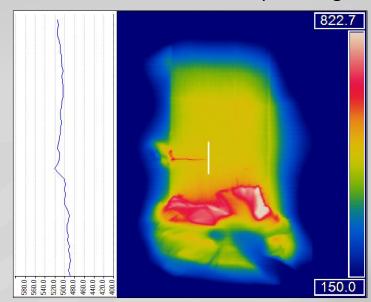




Waspaloy crack length vs. number of cycles: furnace and induction coil comparisons at 650°C, 450MPa and R=0.1.



Ti6246 with crack plane at 500°C. Profile indicates no effect of crack tip heating.



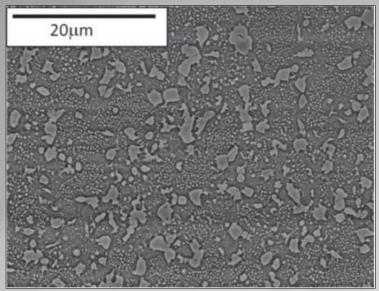


# Fine and Coarse grained RR1000





#### Fine grained RR1000



#### Coarse grained RR1000

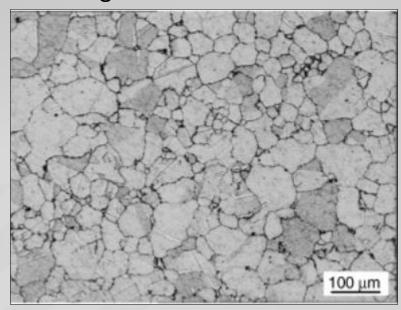


Image courtesy of Li et al, Effects of microstructure on high temperature dwell fatigue crack growth in a coarse grain PM nickel based superalloy, Acta Materialia, Volume 90, 15 May 2015, Pages 355-369

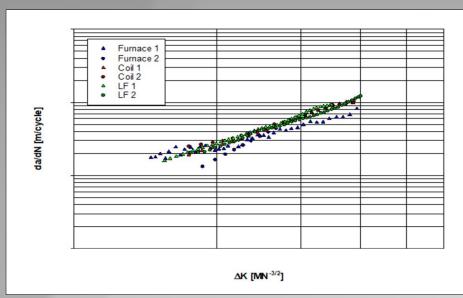


### **Effect of Lamp Furnace vs Induction Coil**





#### **Isothermal**



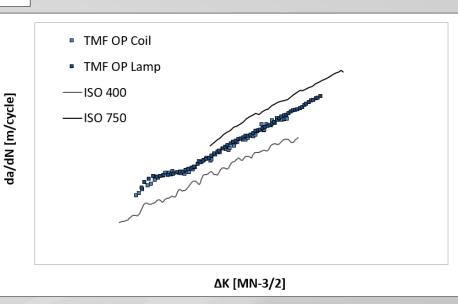
- For out of phase tests no evidence seen of variability in crack growth rate based on method
- Similar results in IP tests but more variability in growth rates overall
- ROLLS

  Dev TMF

  Clean Sky

- ➤ Ti-6246 corner crack specimens tested at the same isothermal conditions, using three different heating methods
- Crack growth rates are consistent across heating methods

#### **Out of Phase TMF**

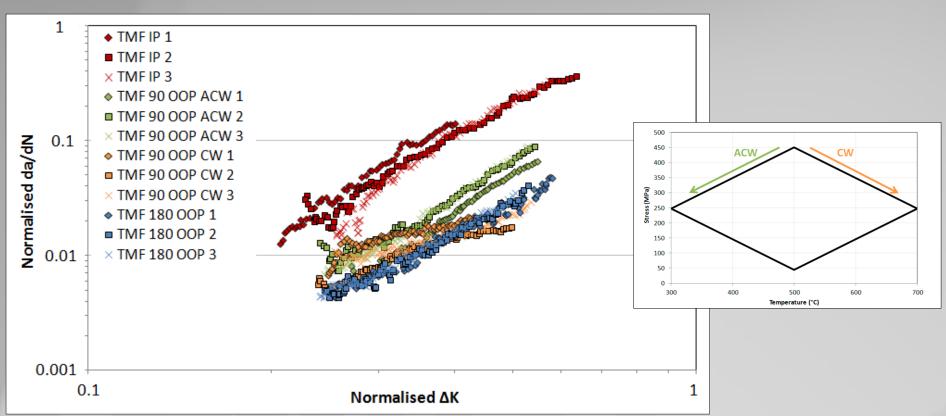


#### Fine Grained RR1000 TMF CP





#### TMF Data 300-700°C



- Strong dependence on phase angle
- Rates tend to approximate temperature at which peak stress occurs

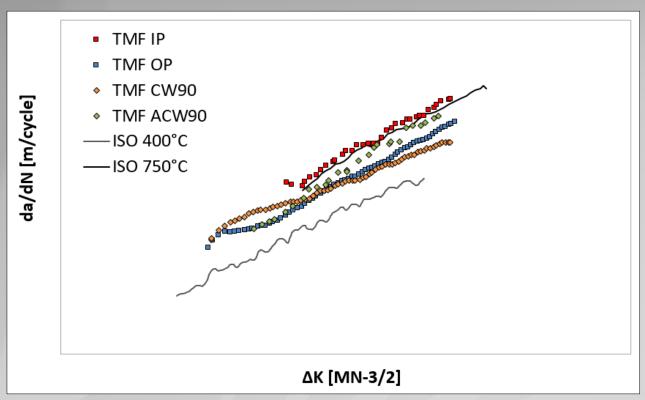


#### **Coarse Grained RR1000 TMF CP**





#### **TMF Data 400-750°C**



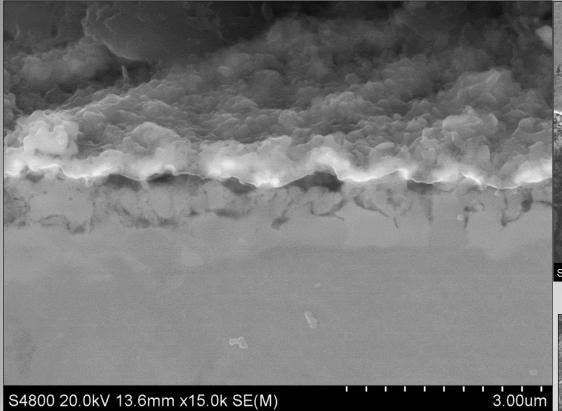
- Phase angle effects are significant in the alloy
- TMF crack growth rates exceed isothermal rates at peak temperature
  - Cycle direction is also important

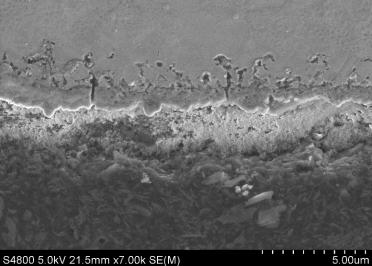


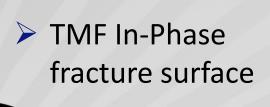
# **Oxidation damage**

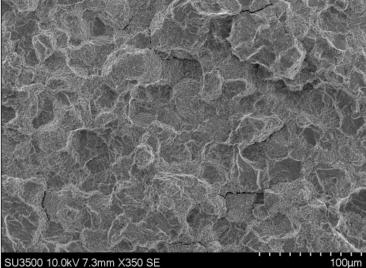












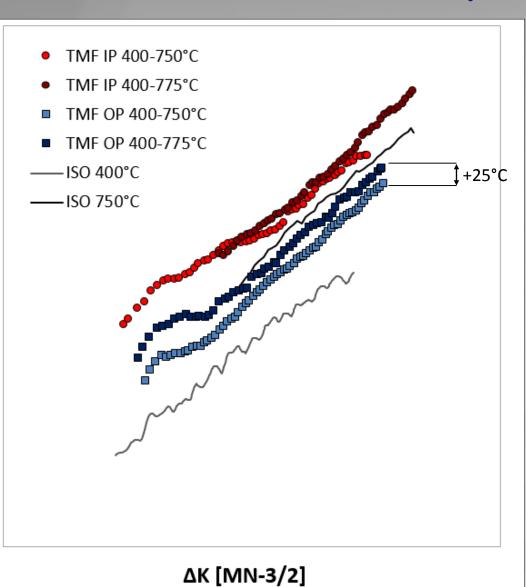
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# **Effects of Peak Temperature**









- TMF CG RR1000
- In phase tests show faster rates than equivalent isothermal tests
- Dependent on cycle time
- In Out of Phase tests increased Tmax influencing TMF CP rates









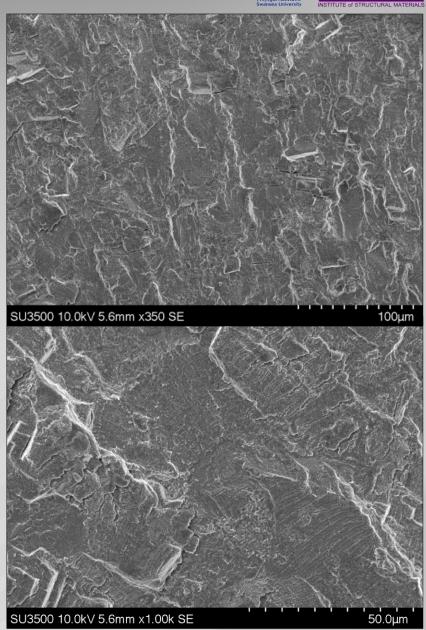
# Mechanisms of crack growth in OP tests





- Transgranular failure indicates minimal effect of oxidation
- Acceleration with increased peak temperature must be due to alternative mechanism
- Compressive stresses at crack tip relieved during high temperature portion of cycle?
- Increased temperature increased relaxation
- More relaxation leads to greater tensile strain and higher increment of crack growth?



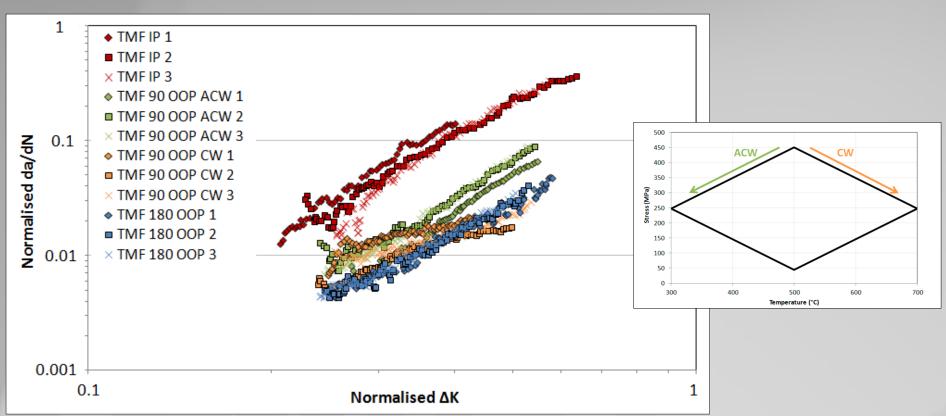


#### Fine Grained RR1000 TMF CP





#### TMF Data 300-700°C



- Strong dependence on phase angle
- Rates tend to approximate temperature at which peak stress occurs



# Results: 90° OOP CW vs. ACW theory





#### > CW:

- Unloaded at high temperatures which oxidises crack tip because there is no crack growth so oxides reach a few grains beyond tip
- Loads the oxidised crack tip so crack grows along oxidised grain boundaries causing more of an intergranular failure

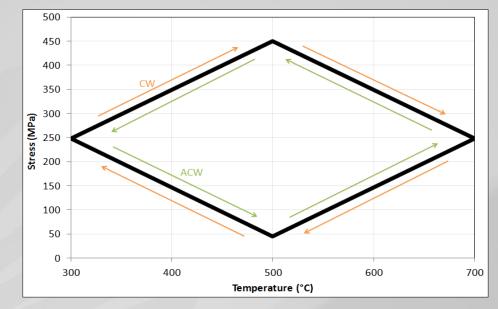
#### > ACW:

Unloaded at low temperatures so there is no/less oxidation

Loaded at higher temperatures so creating new surfaces and preventing oxidation of crack tip causing dynamic transgranular failure i.e. Crack growth faster than oxidation process

#### > Theory to the test:

- 500 second cycle TMF OOP90° ACW test to manipulate mechanism
- Check if oxidation surpasses crack growth rate for intergranular failure

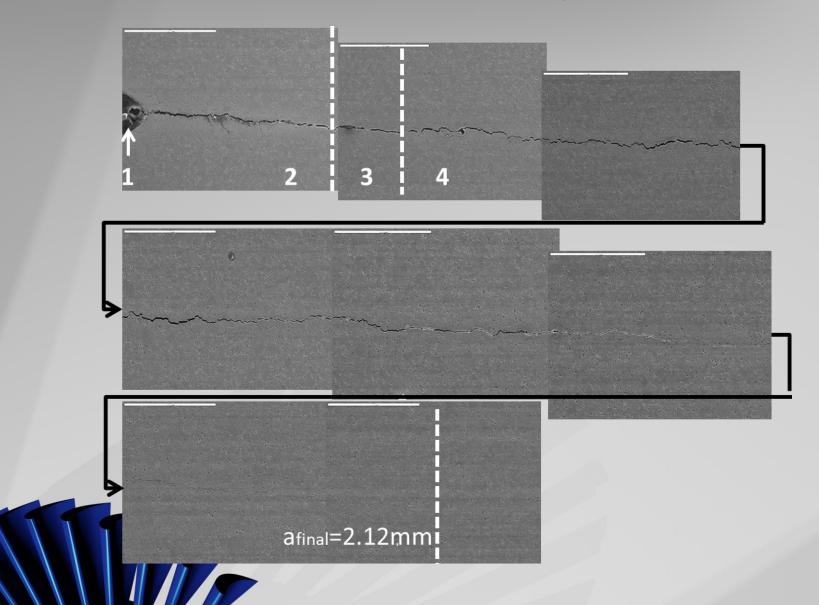




# Results: 90° OOP ACW (80s) Crack Progression



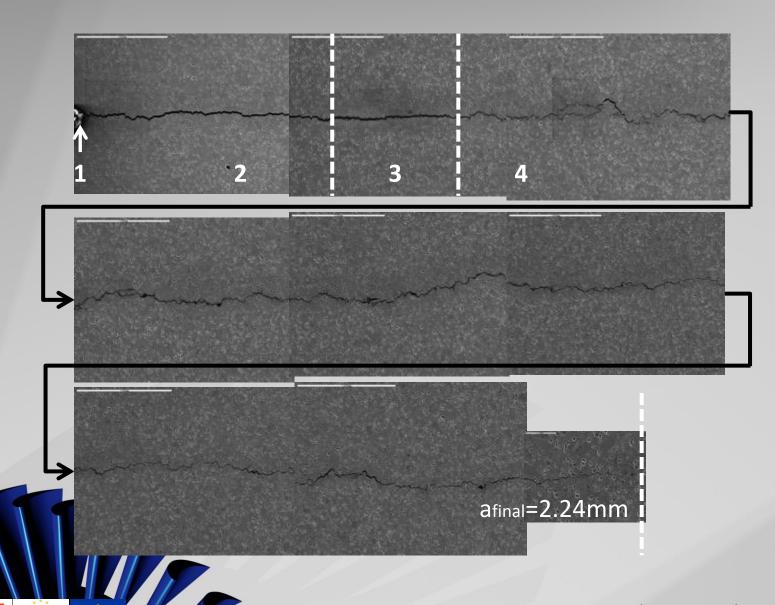




# Results: 90° OOP ACW (500s) Crack Progression



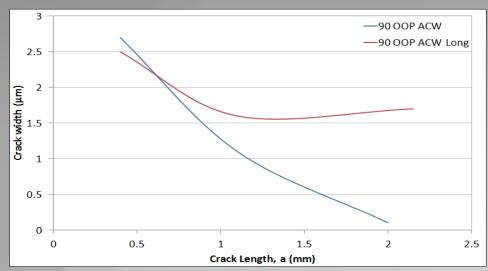




# **Results: Evidence of Crack Tip Blunting**



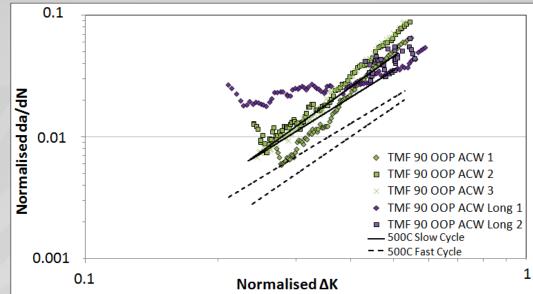




- It is clear that the longer cycle results in a wider crack throughout the test
- Flat gradient of the Paris curve supports this crack tip blunting theory

- ➤ The crack growth is retarded by the reduced stress concentration at the crack tip
- Is this due to oxidation layers on the upper and lower surfaces of the crack?



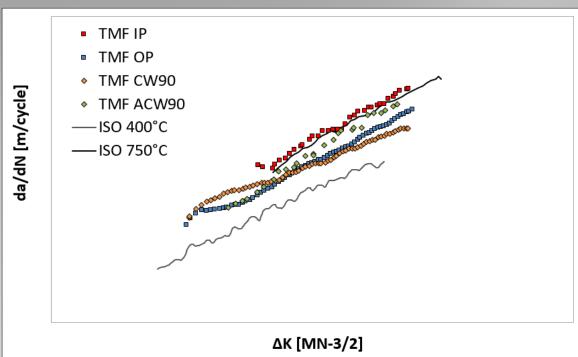


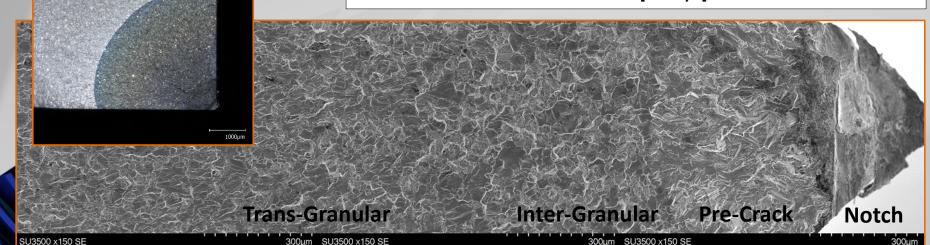
# **CW and ACW Diamond 90 cycles**





- Accelerated crack growth in ACW 90 cycles
- Slow crack growth in CW 90 cycles
- Often a low gradient is seen early in the test (oxidation dominated) giving way to an increased gradient as dynamic failure takes over







#### **Conclusions**

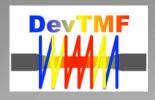




- Crack propagation techniques are developing towards a Code of Practice to enable damage tolerant lifing approaches.
- Significant differences in TMF cycles due to phase angle occur in both fine grain and coarse grain material
- In phase tests dominated by oxidation damage
- Out of phase tests influenced by stress relaxation of compressive stresses around the crack tip. Higher temperatures promote more relaxation which leads to a greater increment of crack growth.
- CW and ACW cycles can be very sensitive to oxidation and rates may depend on cycle times.











# **Acknowledgements**

This project has received funding from the European Union's Horizon 2020 research and innovation programme and Joint Undertaking Clean Sky 2 under grant agreement No 686600.

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# Any Questions?







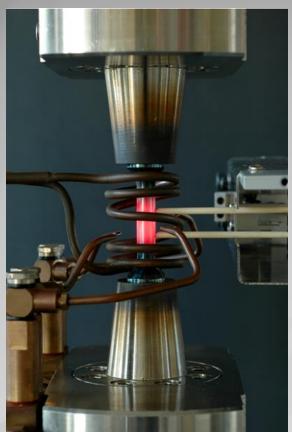


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# **Forced Air Cooling**



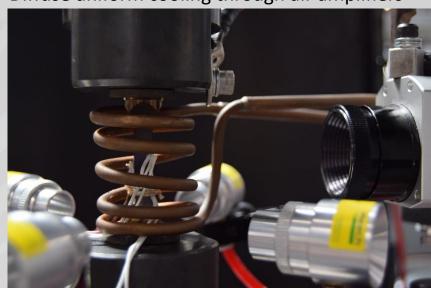




Localised/Focused Cooling



Diffuse uniform cooling through air amplifiers



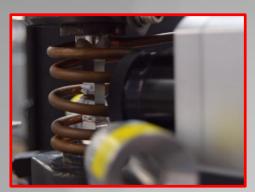
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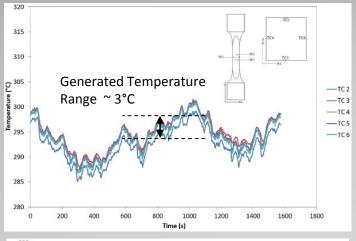


# **Induction Coil Designs**



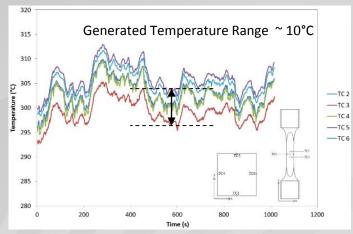


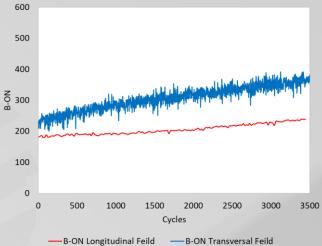






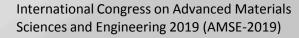












# **Pre-Crack Procedure**











#### **Thermo-Mechanical Fatigue Crack Growth Pre-Cracking**

Stage	Temperature (°C)	Waveform	Frequency (Hz)	Stress (MPa)	Duration (μv)
1	Ambient	Sine	5	600	25
2	Ambient	Sine	5	500	50
3	Ambient	Sine	1	500	75

